

## RAK-33060 Fracture mechanics and fatigue

### 7. Exercise, Failure Assesment Diagram (FAD)

Here the reference to Anderson is to the 3rd edition.

- Anderson, problem 9.6.

A flat plate 1.0 m wide and 50 mm thick which contains a through-thickness crack is loaded in uniaxial tension to  $0.75\sigma_{YS}$ . Plot  $K_r$  and  $S_r$  values on a strip-yield failure assesment diagram, use for various flaw sizes. Estimate the critical flaw size for failure. For the limit lad solution, use the  $P_0$  expression in Table below (Table A9.11 in Anderson). Set  $\sigma_0$  equal to the average of yield and tensile strength. Use the following values  $\sigma_{YS} = 345$  MPa,  $\sigma_{TS} = 448$  MPa,  $E = 207$  GPa,  $K_{mat} = 110$  MPam<sup>1/2</sup>.

$a/W$ :		$n=1$	$n=2$	$n=3$	$n=5$	$n=7$	$n=10$	$n=13$	$n=16$	$n=20$
<b>0.125</b>	$h_1$	2.80	3.57	4.01	4.47	4.65	4.62	4.41	4.13	3.72
	$h_2$	3.53	4.09	4.43	4.74	4.79	4.63	4.33	4.00	3.55
	$h_3$	0.350	0.661	0.997	1.55	2.05	2.56	2.83	2.95	2.92
<b>0.250</b>	$h_1$	2.54	2.97	3.14	3.20	3.11	2.86	2.65	2.47	2.20
	$h_2$	3.10	3.29	3.30	3.15	2.93	2.56	2.29	2.08	1.81
	$h_3$	0.619	1.01	1.35	1.83	2.08	2.19	2.12	2.01	1.79
<b>0.375</b>	$h_1$	2.34	2.53	2.52	2.35	2.17	1.95	1.77	1.61	1.43
	$h_2$	2.71	2.62	2.41	2.03	1.75	1.47	1.28	1.13	0.988
	$h_3$	0.807	1.20	1.43	1.59	1.57	1.43	1.27	1.13	0.994
<b>0.500</b>	$h_1$	2.21	2.20	2.06	1.81	1.63	1.43	1.30	1.17	1.00
	$h_2$	2.34	2.01	1.70	1.30	1.07	0.871	0.757	0.666	0.557
	$h_3$	0.927	1.19	1.26	1.18	1.04	0.867	0.758	0.668	0.560
<b>0.625</b>	$h_1$	2.12	1.91	1.69	1.41	1.22	1.01	0.853	0.712	0.573
	$h_2$	1.97	1.46	1.13	0.785	0.617	0.474	0.383	0.313	0.256
	$h_3$	0.975	1.05	0.970	0.763	0.620	0.478	0.386	0.318	0.273
<b>0.750</b>	$h_1$	2.07	1.71	1.46	1.21	1.08	0.867	0.745	0.646	0.532
	$h_2$	1.55	0.970	0.685	0.452	0.361	0.262	0.216	0.183	0.148
	$h_3$	0.929	0.802	0.642	0.450	0.361	0.263	0.216	0.183	0.149
<b>0.875</b>	$h_1$	2.08	1.57	1.31	1.08	0.972	0.862	0.778	0.715	0.630
	$h_2$	1.03	0.485	0.310	0.196	0.157	0.127	0.109	0.0971	0.0842
	$h_3$	0.730	0.452	0.313	0.198	0.157	0.127	0.109	0.0973	0.0842

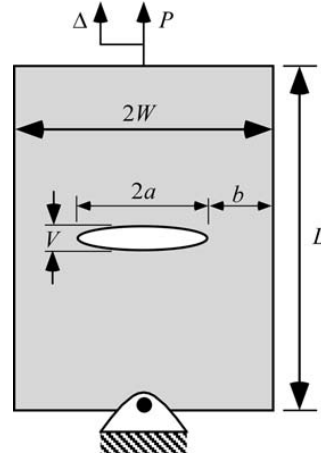
$$J_{pl} = \alpha \varepsilon_o \sigma_o \frac{ba}{W} h_1(a/W, n) \left( \frac{P}{P_o} \right)^{n+1}$$

$$V_p = \alpha \varepsilon_o a h_2(a/W, n) \left( \frac{P}{P_o} \right)^n$$

$$\Delta_{p(c)} = \alpha \varepsilon_o a h_3(a/W, n) \left( \frac{P}{P_o} \right)^n$$

$$P_o = 2Bb\sigma_o$$

$$\Delta_{p(nc)} = \alpha \varepsilon_o L \left( \frac{P}{2BW\sigma_o} \right)^n$$



2. Anderson, problem 9.7.

Consider a single edge-notched tensile panel with  $W = 1$  m,  $B = 25$  mm, and  $a = 125$  mm. Plot the  $J$  results in terms of a failure assesment diagram.

- (a) Compare the FAD curve determined by normalizing the  $x$  axis with  $P/P_0$  to the FAD curve that is normalized by  $\sigma_{\text{ref}}/\sigma_{YS}$  where the reference stress is defined in Section 9.4.4. Neglect the Irwin plastic zone correction.
- (b) Repeat part (a), but include the Irwin plastic zone estimate in the first term of the  $J$  estimation.

$a/W:$		$n = 1$	$n = 2$	$n = 3$	$n = 5$	$n = 7$	$n = 10$	$n = 13$	$n = 16$	$n = 20$
<b>0.125</b>	$h_1$	3.58	4.55	5.06	5.30	4.96	4.14	3.29	2.60	1.92
	$h_2$	5.15	5.43	6.05	6.01	5.47	4.46	3.48	2.74	2.02
	$h_3$	26.1	21.6	18.0	12.7	9.24	5.98	3.94	2.72	2.0
<b>0.250</b>	$h_1$	3.14	3.26	2.92	2.12	1.53	0.960	0.615	0.400	0.230
	$h_2$	4.67	4.30	3.70	2.53	1.76	1.05	0.656	0.419	0.237
	$h_3$	10.1	6.49	4.36	2.19	1.24	0.630	0.362	0.224	0.123
<b>0.375</b>	$h_1$	2.88	2.37	1.94	1.37	1.01	0.677	0.474	0.342	0.226
	$h_2$	4.47	3.43	2.63	1.69	1.18	0.762	0.524	0.372	0.244
	$h_3$	5.05	2.65	1.60	0.812	0.525	0.328	0.223	0.157	0.102
<b>0.500</b>	$h_1$	2.46	1.67	1.25	0.776	0.510	0.286	0.164	0.0956	0.0469
	$h_2$	4.37	2.73	1.91	1.09	0.694	0.380	0.216	0.124	0.0607
	$h_3$	3.10	1.43	0.871	0.461	0.286	0.155	0.088	0.0506	0.0247
<b>0.625</b>	$h_1$	2.07	1.41	1.105	0.755	0.551	0.363	0.248	0.172	0.107
	$h_2$	4.30	2.55	1.84	1.16	0.816	0.523	0.353	0.242	0.150
	$h_3$	2.27	1.13	0.771	0.478	0.336	0.215	0.146	0.100	0.0616
<b>0.750</b>	$h_1$	1.70	1.14	0.910	0.624	0.447	0.280	0.181	0.118	0.0670
	$h_2$	4.24	2.47	1.81	1.15	0.798	0.490	0.314	0.203	0.115
	$h_3$	1.98	1.09	0.784	0.494	0.344	0.211	0.136	0.0581	0.0496
<b>0.875</b>	$h_1$	1.38	1.11	0.962	0.792	0.677	0.574			
	$h_2$	4.22	2.68	2.08	1.54	1.27	1.04			
	$h_3$	1.97	1.25	0.969	0.716	0.591	0.483			

$$J_{pl} = \alpha \varepsilon_o \sigma_o \frac{ba}{W} h_1(a/W, n) \left( \frac{P}{P_o} \right)^{n+1}$$

$$V_p = \alpha \varepsilon_o a h_2(a/W, n) \left( \frac{P}{P_o} \right)^n$$

$$\Delta_{p(c)} = \alpha \varepsilon_o a h_3(a/W, n) \left( \frac{P}{P_o} \right)^n$$

$$P_o = 1.072 \eta B b \sigma_o$$

where

$$\eta = \sqrt{1 + \left( \frac{a}{b} \right)^2} - \frac{a}{b}$$

$$\Delta_{p(nc)} = \alpha \varepsilon_o L \left( \frac{P}{2BW\sigma_o} \right)^n$$

